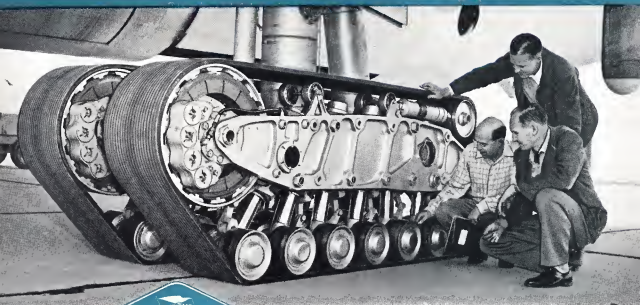


AVIATION WEEK

AUG. 21, 1950

A MCGRAW-HILL PUBLICATION



It makes the B-36 light on its feet

THIS caterpillar-type track gear that permits the giant B-36 to operate from quickly prepared landing strips was developed jointly by Convair and Goodyear engineers. Goodyear experience produced the

rugged endless rubber tracks, brakes and brake bogies used in this wide "footprint" gear that safely spreads the B-36's weight.

Goodyear, Aviation Products Division
Akron 16, Ohio or Los Angeles 54, Calif.

MORE AIRCRAFT LAND ON GOODYEAR TIRES, TUBES, WHEELS
THAN ON ANY OTHER KIND

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FUEL GAGE
for your
AIRPLANE**

Honeywell's standard
fuel gauge fits both
military and civil
aircraft fuel gauges.

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include the remote
indicator and the fuel
quantity indicator.



Reliability, the quality of being persistently and dependably right, is the ultimate requirement of an airplane fuel gauge. Reliability is accurately measuring fuel available for engines at all times. Reliability is functioning day-in and day-out without troublesome maintenance.

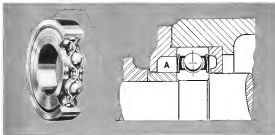
To meet this prime requirement, every facility of the Honeywell engineering, research and manufacturing staff has been brought to bear on perfecting a fuel gauge which applies sound electronic principles to the problem of fuel quantity indication and incorporates rugged lightweight construction for optimum performance under adverse conditions, and with minimum maintenance cost.

Honeywell Fuel Gauges are built to a plant demand exclusively to the manufacture of electronic control and measuring devices for the aircraft industry. Back of this highly specialized manufacturing organization is a complete program of testing and re-evaluation of not only the gauge itself, but fuel characteristics and installation problems as well.

For five years Honeywell Fuel Gauge installations have proved the rightness of the Honeywell system. Now new component designs increase its adaptability... make it right for your airplane because it is engineered right, built right, and installed right. Minneapolis-Honeywell, Minneapolis 6, Minnesota, 16 Canada: Toronto, Ontario 17, Ontario.

**MINNEAPOLIS
Honeywell**
AERONAUTICAL CONTROLS

Long Time No See!



Most bearing users want bearings that will give them years of dependable service without fussing over lubrication, adjustments, etc.—bearings that can be forgotten for long periods and no harm done.

For instance, take a standard width New Departure ball bearing, shrouded on both sides—what it is shown as shown above—fill space "A" full of the recommended

grease and under anything like normal conditions that bearing will run sweet and smooth for years without attention of any kind. You can't even see it. It's not underground; and in an electric motor for example it's good for any position from horizontal to vertical. Any New Departure representative will be glad to give you details.

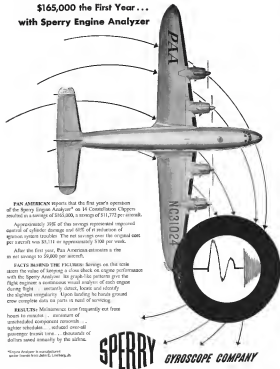
Nothing Rolls Like a Ball

NEW DEPARTURE BALL BEARINGS

NEW DEPARTURE DIVISION OF GENERAL MOTORS BOSTON, CONNECTICUT

Pan American saves

**\$165,000 the First Year ...
with Sperry Engine Analyzer**



PAN AMERICAN reports that the first year's operation of the Sperry Engine Analyzer* on 14 Constellation Clippers resulted in a savings of \$165,000, a savings of \$11,772 per aircraft.

Approximately 35% of this savings represented improved control of cylinder damage and 40% of it reduction of ignition system troubles. The net savings over the original cost per aircraft was \$1,111 or approximately \$100 per week.

After the first year, Pan American estimates it is the net savings to \$9,000 per aircraft.

FACTS BEHIND THE FIGURES: Savings on this scale stem the value of keeping a close check on engine performance with the Sperry Analyzer. Its graph-like patterns give the flight engineer a continuous visual analysis of each engine during flight... instantly detect, locate and identify the slightest irregularity. Upon landing he hands ground crew complete data as parts in need of servicing.

RESULTS: Maintenance time frequently cut from hours to minutes... reduction of unscheduled component removals... tighter schedules... reduced over-all passenger transit time... thousands of dollars saved annually by the airline.

*Engine Analyzer is manufactured
and sold by Sperry Gyroscope Co., Inc., New York, N.Y.

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News Picture Highlights . . .



PHONE TO FLY

A modified F40 fitted with a voice-projection cockpit has started flight tests under supervision of Stucky Aviation Corp., Buffalo, N. Y. Red at Nylon setting conforms to body contour.



MORE POWER TO THE RAIDER

This Northrop G-125 transport has more powerful 1575 hp. Wright R-1520 101 engines in place of the standard 1200 hp. R-1073-9's. Bulk orders will be received at Eglin Field.



SABREE'S AUXILIARY WING TANK

New design underlying fuel tank down installed under Nauck American F40 wing is not droppable, carries about 125 gal. Not fuel line with negative differential.



A NEW MIDGET RACER

This 55 hp. Continental-powered racer features an overridable wing, permit- ting direct attachment of leading gear to wing. It was designed by Neil Leving (in cockpit), head of Wayne School of Aeronautics, Detroit.



ARGENTINE ALL-WING EXPERIMENT

The Argentine Institute Aeronautica is experimenting with this rubber glider developed for the Argentine to learn to carry out research on all-wing aircraft. The 1-1/2 ft. 54 Class (also Star Ray).

with two individual tandem cockpits. Span is 59 ft., overall length 14.45 ft., empty weight 605 lb. Wing is swept about 22 deg. at quarter-chord. Area is 224 sq. ft.



PHOTO BY COURTESY OF GENERAL AVIATION

A Good Sign to Fly to...



As the principal airport serving Madrid, Barajas is fully equipped to accommodate international airlines of all types and sizes. Esso Aviation Products and services are regularly relied on here as elsewhere along the airways of the world. Constant research and development on Esso Aviation Products keep pace and even anticipate the constantly changing requirements of modern aviation. The Esso winged oval symbolizes products of uniform, controlled quality backed by more than 60 years of aviation experience.

*All Barajas Airport and throughout Spain, the markets of Esso Aviation Products is Shellford Oil Company of Spain, S.A.

ESSO EXPORT CORPORATION, AVIATION DEPARTMENT, 25 BROAD STREET, NEW YORK 4, N.Y.

WHO'S WHERE

In the Front Office

J. D. Conner has become the vice president finance, aviation division, and assistant director of Pacific Northwest Corp. He held executive positions with Conquest and several large custom corporations prior to his present assignment. In what has been his last position, TAC-Berkeley he has made vice president of the company's custom division with headquarters at Los Angeles, N. J. Earl Palmer has been promoted to the president.

James D. Aiken has been appointed vice president general manager of Ponderosa Products, Inc., N. J. Effusive equipment sales. He was previously assistant to the president and has also served as assistant sales manager, aircraft plant inspection unit, quality manager and cost reduction engineer.

Glenn G. Whitaker was elected a director of the Wm. W. Brinkley Co., Ltd., an addition to his duties as vice president field engineering.

Changes

Reg. Geo. William G. Smith has been named chief of the Marine Air Transport Service, Ltd. Winston-Salem, N. C. **William B. Brown** has been made manager of Grand Central Aircraft Co.'s new Turbine plant.

Jack S. Baker has been appointed assistant manager of Billy Mitchell Field, Md. under.

Among the Massachusetts-J. David Wright has been named assistant manager of General Electric's Industrial division at Springfield, N. Y. **Frederic M. Roberts** is GE's new manager of Industrial Engineering division, and **Leonard A. Krasinski** has become manager of engineering of the Industrial Engineering division.

John R. Griffin has joined the Pratt & Whitney Chemical division as aviation consultant. **Capt. John Murphy** has been appointed head of the research, development, and test division of the Lockheed Aircraft Division. **Arthur F. Palmer** has joined Jack S. Baker's Pratt & Whitney Chemical division.

Two changes in Montgomery Aircraft's administrative engineering staff. **Walter E. Dwyer** has taken over the new position of assistant to the manager of engineering and manufacturing, and **William H. Hurd** is coming project engineer to one of the company's major engine power plant projects.

William R. Baker has been named vice president and general manager of Hughes Aircraft Co. **Robert H. Gode** has been made assistant traffic manager of Lockheed Aircraft. **George S. Hildebrand** has been appointed supervisor, mechanical lab of General Electric.

Robert Choddy has joined Comair Aircraft as director of sales promotion and public relations.

INDUSTRY OBSERVER

►Transport potentialities of the Conquest XP5-Y1 turboprop Navy flying boat, plus its steadily improving showing in economic test flights, may result in an initial production order for ten planes. The big plane's last test flight went over eight hours, more than doubling any previous flight.

►Turboprop versions of the two heavy transports in the Air Force program—the Douglas C-124A and the Boeing C-97A—are probable future developments. Now Air Force interest in the Pratt & Whitney turboprop T-34 engine (company designates FT-2) indicates it may be headed for these two transport assignments.

►Plans to convert one of the two McDonnell F-8B experimental jet penetration fighters to turboprop power will retain the plane's present twin Westinghouse J34 turboprop engines, and add a third propeller—on the Allison T-35—in the nose, with a 104-in. diameter eight-blade Curtiss propeller. Armor will be stripped from the plane to permit the additional propeller installation. Project aims to make first flight test of new three-bladed propellers of two engines, or possibly superior capabilities, in an airplane with design capabilities of Mach 1 or better.

►The Canadian night fighter CF-106 Conquest last week flew from Toronto to Montreal in 31 min. 30 sec. for an average speed of 634 mph, setting a new turbo-jet record.

►General Aircraft's Libertyway has developed a small variable-pitch propeller engine to power guided missiles and helicopters. It is evolved with quick response and performance over the other propeller designs for the General V-3 propeller. Reed valves in the V-3 type engines were a source of flow loss and were subject to severe fatigue at high temperatures, thus limiting a true limited service life. The performance drop is obtained by modifying the shape of the duct and elimination of the valves.

►A Piper Cub used in recent stall warning tests, is giving out in the CAA region to demonstrate results of the tests which were conducted by National Research Council for CAA. Plane was equipped with a stall warning indicator, an angle of attack indicator and a radio altimeter.

►British Aerospace Mark II transport, despite its large size, was making landing and takeoff runs as short as 1200 to 1400 yd. in recent flight tests at London Airport. Although it was not operating at full gross weight (300,000 lb.), British engineers believe the tests show that it could operate at full gross loads without any further substantial airport.

►Use of boundary-layer control devices can reduce the total landing distance for a turbojet engine airplane from 25 percent to 40 percent below what it would otherwise be, NACA analysts have reported. Calculations used various conditions, including wing spans of 25 to 180 ft. and engine power from 100 to 1200 hp, for the study.

►An experimental turbo-propeller version of the Boeing B-47 is now in a planning stage. Presumably it is intended to give the USAF some indication of a new possible jet with a turbo-propeller version of the Boeing XB-52 eight-jet experimental bomber, which later became the turbojet version now under construction.

►The Sperry Zero Radar for British Overseas Airways' Conquest and other planes (Aeronautics Week, Aug. 14), probably will be made or assembled in Britain to avoid payment of duties for the equipment. If it proves practical to produce the Radar in England, it should increase Sperry's chances to sell the instrument to the Royal Air Force, which already has expressed interest. RAF is increasing its fleet of night fighters, for which the Zero Radar apparently is of special value, judging from USAF's orders for the instrument for F-96, F-98 and F-104 fighters.

►The third SAAB Scania, twin-engine transport, of an order for six by VASP Airlines, has left Sweden for delivery to the Brazilian airline. The VASP planes are arranged for 32 passengers and are being put into service between Rio de Janeiro and Sao Paulo, Brazil's heaviest-traveled route.

Domestic reports are getting at least as much of the new attention being turned to the hydrocarbon end of natural gas.

A 12-litre pressure hold system for 91 and 100 octane fuel has recently got into limited service in the hangar area at the Philadelphia International Airport, according to Esso Standard Oil Co. The system is available for Wings, Inc., Panavia Helicopters and the USAF.

Renewable concrete blocks containing sand-filled trenches permit rapid and inexpensive installation of additional piping from bulk storage to hydrants when expansion is required.

Within two years Baco intended experts to install a separate hydrant system in front of the proposed terminal building of the same airport to service scheduled airline departures. Also in view is a third system for the existing cargo and freight area. Agents of Houston and Fort Worth, Tex., will soon have consultations.

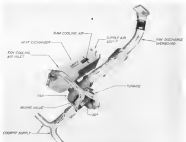
The company is considering an installation at Friendship Airport, Baltimore, Md., but is waiting data from the airlines serving the field to determine whether the fuel demands will be large enough.

Esso and airports where the fueling system has been in operation are Teterboro and Atlantic City, N. J.; Wheeling, W. Va., and Wallaceton, Pa.

- **Low costs:** Many Exports rate the initial cost of installing a hybrid trash-drying combination as its export is not unreasonably high. This, coupled with the low maintenance cost of few moving parts, makes the company's products desirable to the owner for low total cost of ownership.

Shell 420, on the other hand, points to trucks in leaving the flexibility needed for part-time day operations. It has one pump (usually) gal in service (4300 gal. trucks which can service two- and four-stage plants at rates up to 400 gpm. from two bores). Of the two-side rather than the trunk type, they can be used in pairs. Shell notes that truck-trucks are held in much as 5000 gal., but adds that they require highly skilled operators and are difficult to handle.

Also Expert has supplemental track, source charts available at airports where the lockout system is in use.



Unit Air-Cools Fighter Cabins

Palladium production deliveries are expected to rise as manufacturing needs, new development of Brazilian Stradale in the North American (NA) market, and the new American (NA) market, are scheduled for later this year, the company announced.

The 20th, west delivers a 120-mph hammer to the fighter's cockpit, changing the 60 cu ft of air every 15 seconds the engine sees.

The pilot streamers are drawn out of the compressor by controlling a thermostat which operates a mixing valve blending air bled from the engine and reduced to 715 deg. F. with air having passed through the expansion turbine.

The manufacturers pointed out that cooling; high-speed airflow cools to a minimal level but cannot prevent the bacteria with the atmosphere, electrical equipment in the cockpit and solar modules have heated inside temperatures to as high as 160 degrees F.

a special subscription system for the

60,000 rpm, 3-in. turbine wheel, which is the best of the rest, has enabled our production model to operate over 1000 hr without maintenance of any kind, the few states, while five other units on our machine, the same time.

Mike Martin, **Harn Standard** general manager, says that development of the air conditioning unit has paralleled his company's intensive work of recent years on high speed propellers and turboprop mechanisms. From a product standpoint, he continued, the unit will utilize the basic machine tools and manufacturing skill available at Harn Standard without heavy investments in tooling.

Ignition Analyzers

Bentley Systems, Inc., N. Y., is said that the Navy is installing its system analysis in 19 Martin P-450s. The new Bentley contract is estimated at \$50,000.

NEW AVIATION PRODUCTS



After studying 57 different ads, engineers realized the ad they were looking for simply didn't exist. So they created a new one and perfected it into the top grades now on the market.

Advantages derived from the new unit
shown below:

- They don't "creep" or spread over parts of the instrument when they're not wanted. They "stay put" for a long time when applied in minute quantities to small protrusions or protuberances.

- They were evaluated and reported parenting them to be used for long periods with a minimum of guidance or direction.

Because of its lower viscosity point, Merox Burning Oil is recommended over the "Special" when it is desired to get the longest possible use out of the oil before a new application



Navigation Computer

A navigation computer for private pilots which may also be used earlier with time and fuel consumption figures to simplify and speed computations is being marketed by the Glenside Co., Ft. Worth, Texas.

The firm says use of the device is not confined to private pilots. According to the company, the Air View has been subject and currently is receiving, with the new instrument.

The "Quickpak" consists of a firm rigid plastic scale on which is printed a 180-deg. protractor and a speed scale calibrated from 60 to 300 mph. The elastic time band is attached to one end of the scale.

Brinkley, this is the way the computer is used to find the length of time it takes at a given speed to fly between two points on a map.

- Rigid scale is placed at viewing position on road
- Plastic laser band attached to scale is stretched along centre line until an index arrow on the band is opposite the plane's speed on the rigid scale

* Time markings then will show, at that point on the tape opposite the cluster's true point on the map, the time is spread to fly the course. Markings on land also will show time, it will take to reach various fixed points.

Flight required to fly 1 gram course, at a given speed is computed as usual by the basic formula. Again, the rapid scale is placed at the starting point on the map and the cloth band stretched until it fits on it, corresponding to known fuel consumption per hour, is opposite the known speed of the plane. The number of gallons required for the flight is shown on the elastic band scale adjacent to destination point on map.

ALSO ON THE MARKET

High gain becomes intense for gun of about 3 db, can be used for converging cables between local outputs, control tower to aircraft on ground, various other uses, etc. Makani says it fits with use of low cost radio equipment. Address: Wavelength Associates, Inc., 115 Crescent Rd., North Attle, Mass.

A complete feedstock type-report feeding system has been designed and patented by Harg Mesta Systems, Inc. The system uses mobile dispensing units carrying hoses from ground feedstock to aircraft. Patents also include pumps and water recovery units. Address: Harg, Pa.

A shock-absorbing swirl, recently developed by J. & M. Barrett Co., can be fitted to any standard drill indicator with a solid contact point to construct a unit: a shock-absorbing instrument. Internal spring in swirl does not interfere with normal functioning of indicator, resists shocks and blows, absorbs down strokes to equalize use of instrument. Thread size is No. 4-48. Address: Mfg. Div.

Another lightweight headset to add to the arsenal is the "Raptor" headset. This unit has a quick-adjusting cap piece, making it adaptable to all ears, making it suitable to make it weigh less than 1 oz. It's supplied with a 6 ft cord and jack. Address: Dtec Rump Co., P. O. Box 4178, St. Wust, Texas.

Line of metal parts washers with solvent capacities ranging from 5 to 75 gal are offered by Flo-Tite Heat Control Co. Metal washer tank is equipped with light fitting cover, handle to suspend parts in solvent, filter screen to clean solvent and impeller stir for 400 to 500 rpm. Large units have pumps, hose, and controls for mixing. Models are suitable for a slight. **Glezer 9125** is unit Ave., Cleveland 5, Ohio.



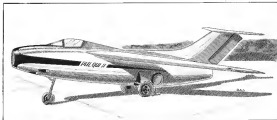
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AERONAUTICAL ENGINEERING



ARGENTINA'S latest turbojet aircraft is shown here in sketch based on Buenos Aires newspaper photo. Recognition features in text box.

Pulqui II: Newest Argentine Jet Plane

Conventional layout of subsonic fighter shows German design features.

By David A. Anderson

Fast details of the Argentine Republic's newest jet fighter, Pulqui II, are becoming available despite the lack of official information on the plane.

Plans first on June 17 by Capt. Edelberto Weiss, Pulqui (Arrow) II is the second Argentine jet. Two days later, on its first flight, the craft ground-looped on landing. It now is under going repairs.

Design credit for the airplane goes to Dr. Kurt Tank, late technical director of Focke-Wulf Flugzeugbau GmbH. Builder was the Instituto Aerodinamico de Canada, government aircraft factory. Argentina's first jet, Pulqui I, was designed by a French engineer, Renee Devostine, and was flown about three years ago. It also was built by the Institute.

Flight Data—Argentine Weiss's Buenos Aires correspondent was able to get little information on the Pulqui II beyond the photo and story which appeared June 15 in the official morning newspaper, *Democracia*.

The newspaper said that the plane took off for its maiden flight at Corrientes with Capt. Weiss flying. A 400-meter (1,300 ft) ground loop put it into the air, and from then on during the half-hour test flight the plane performed

well at altitudes up to 10,000 meters (just short of 33,000 ft).

Democracia claimed that speeds of 1,000 mph (821 mph) were reached, but the Buenos Aires *Atrevida's* office in Buenos Aires felt that 450 mph was a more reasonable figure.

Observers there were convinced that Weiss, in common with other Southflight pilots, did not touch the engine.

Second Flight—On June 19 it was flown again, this time by Capt. Ludwig Rejovsky, late of the Luftwaffe and once a Focke-Wulf test pilot.

Official story on Rejovsky's flight was that an accident was sustained on landing.

Unofficial sources said the damage to the aircraft was confined to a washed-out landing gear and a twisted wing because of a ground loop on landing.

Pictorial Analysis—The sketch of Pulqui II reproduced here is from the photo in *Democracia*.

At first glance, the airplane appears to be a conventional jet layout, with no canards in the nose. A shoulder-height wing is mounted on a bit fuselage. The cockpit gear of extremely narrow track (which may have contributed to Rejovsky's accident) is small. A Rolls-Royce Nene powers the craft.

Chief recognition feature of the plane is the T-tail, with the horizontal stabilizer mounted at the upper extreme of the vertical fin.

A second glance at the picture shows some of the major points of design interest in Pulqui II's layout.

From the shadow cast on the ground, it would seem that the wing is of constant chord, probably with the now-scrapped 35-deg sweepback. The horizontal tail is tapered in about 2:1.

(Both these features were as a Focke-Wulf proposal made in early 1945 for a jet interceptor. The proposal aircraft was designated the Tu-157, taking its designation from Turkey's name.)

Nonconventional leading gear also was featured on the FW proposal, and seems to have been picked up for Argentine use. It appears that the gear is of the single leg type, followed by a Geneva aircraft design and that it retracts around a slanted stem, without folding, to fit into a fuselage well.

The T-tail was a conspicuous item on the Tu-157 proposal, with a notable difference in the span chord ratio at the vertical fin. The earlier Geneva proposal showed a vertical surface about three chords high, Pulqui II's is a single flat surface. It's a mile bet that the latter choice was made because of the possibility of tail flutter.

Armament is probably four cannons.

Mixed Blood—Democracia calls the new jet an "all-Argentine" fighter. But the fact remains that it was designed by one of Germany's best engineers, not flown by a former Luftwaffe pilot, and has a British engine.

In sum, Argentina's latest fighter appears to be based on a design that is conventional and five years old, and likely will turn in a performance that is creditable, but not amazing.



MILLING MACHINE at Boeing: North plant can handle 65 ft. long parts; automatic jigs will replace three men's jobs.



Largest Miller

Boeing's new machine, leased from Onsrud, said to be world's biggest.

Boeing Airplane Co. has installed at its Seattle, Wash., plant what it claims is the largest milling machine in the world. Manufactured by Onsrud, the latest machine is capable of handling a 65 ft. long wing spar (Boeing's present capacity is a 35 ft. length).

Unique feature of the machine is its air feed lubricating system, engineered by the Sun Oil Co. This system includes a visual device by which the operator can check the oil flow through the 12 lubricating lines. In the event of line clogging, the operator can either blow the line clean manually, or let the machine blow it clean when the oil reaches a preset level.

The milling machine can have two cutters on horizontal, two on vertical at the same time. It operates automatically on preset cutting operations and has an automatic air cylinder for reaming and reworking outer leads on the miller.

► **Fast Replacement**—Further speeding of production should be possible with the use of a new joggling machine. This speed-purpose tool was designed and built at Boeing to perform operations previously done on press jacks. It is reported that the work at these presses will be handled by one machine.

Indication of the capacity is Boeing's figure of 360 longions joggled per hour, either rolled or extruded sections as long as 7 ft. 7 in. in its length.

The joggling machine is completely automatic, hydraulically powered, electrically controlled and fitted with hand-wheel controls for setting the joggle length, depth and forward angle of joggling.

It is designed for either hot or cold joggling with thermostatically controlled superheated steam circulating through the dieblock.

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"Philippine Air Lines was one of the first international air carriers to use Champion Spark Plugs on overseas schedules. This early confidence was upheld by outstanding performance and, as a result, we have installed Champion K375-1 Spark Plugs in all four-engine planes. Champions are also used extensively in our domestic equipment. They have played an important role in ensuring the dependability and efficiency of P.A.L. planes on scheduled flights over almost 25,000 miles of international air routes."



EDWARD T. BOLTON
Vice President-Secretary

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(Look to the CHAMPION BOLL GATE, a Boxy Wizard's test sponsored every Friday night, over the ABC network)

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terminations per hour.

This electrically operated AMP machine will terminate leads as fast as an operator can insert them! Special CORROSION PROOFED terminals feed into the clamping jaws in strip form—each automatic stroke yielding a superior vibration proof connection. When AMP PRE-INSULATED terminals are used, reworkings are made in labor, time and material since additional taps or taping need not be stocked or applied. There is a wide variety of terminal types and sizes available.

Many users prefer AMP identification bands with these machines to give permanent code numbers to easier to component leads.

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AMP DEVELOPMENTS IN SOLDERLESS WIRING

New corrosion proofed terminals will be shown for the first time! Special coding identifies their type and size.



AMP Products Division
U.S. Patent Office

AIRCRAFT-MARINE PRODUCTS INC.
1328 North Fourth Street, Harrisburg, Pa.



SUBMERGED LAUNCHER for B-29, spin-stabilized rockets replace extremely long projectiles, deliver payloads of 5000 lbs.

Service Demands Cause Design Changes

Firepower and range
increases are achieved
by external appendages.

Increased range and more offensive firepower for military aircraft are creating new demands of the services.

To meet these demands, field and factory modifications or additions are quickly engineered. But since the aircraft's internal structure has generally been frozen by then, and production is well under way, the last tendency is to consider where, on the outside of the airplane, the particular solution to the problem can be hung.

That is the lesson for the clean, production airplane of today becoming the service "Clematis" type" bomber.

Here are some examples of the most recent cases of airplane retooling and modifications to meet combat needs.

► **Fourier** (Panda)—Naval acquisition of the Grumman F4F series is from 1940 aircraft carrier. This offensive weapon is augmented, in the F4F-28, by a built-in 5-in. rockets and a pair of 1000-lb. bombs.

The 14-light picture of the Fourier shows only the rocket armament. This particular ship is an F4F-3, being used for flight test work.

But more intriguing than the second arrangement is the appearance of a set of fuelage as booms, visible in the picture as a group of perforations in the bottom of the fuselage, just ahead of the landing edge of the wing.

► **Siber** (Sawyer)—North American's F80 fighters are sporting a new external gas tank to give them increased range.

The last 14-light photo shows the somewhat unconventional shape of the



WING-MOUNTED ROCKETS on P-40 are light tested on today's P-40 fighters.



EXTRA TANKAGE is supplied on the F80 Siber from two non-droppable "banana" tanks.



Every Facility to Produce PRECISION GROUND PARTS

Attention to angle, line and accurately centered, true machines
milling machines, horizontal or vertical modern and complete grinding
equipment of every type multiple and single-pointed drill presses
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tank, which has led to its application of
"bore" design.

From this and other photographs, it
would appear that the tanks are aligned
in cross-section, with the major axis
parallel to the wing chord plane. Tank
extension is kept up at the trailing
edge, probably to match the airfoil
pattern around the wing.

In keeping with the trend of the
times, the tank carries a horizontal line
for stabilization during performance.

North American claims that the new
tanks have very little effect on performance,
in fact, they are intended to be
kept on during combat. Further, NAA
says that the tanks do not limit the disc
speed of the Sabre because of buffeting.
It was trouble with buffeting caused by
either the tanks on the P-56 that led
to the design of these bores tanks.

All Sabres coming off the production
line will be equipped with the new
tanks, and earlier models will be service
fitted to make use of them.

Skyrocket Speed—Picture of the
breakdown testing. Douglas AD-100, which
shows it carrying everything else that
could conceivably be carried aloft,
have prompted questions about the ad-
vantages of retracting the landing gear to
reduce the drag.

(Through an fault of the Douglas
designers, the Skyrocket has been in-
correctly misnamed in a number of ways,
and has been featured with all manner
of drag production. That it has succeeded
in a multi-purpose thrust is a tribute to
the engine.)

Possibly perhaps by an early pro-
cess of the AD-100, and others like it, the
Navy has come up with a general mod-
ification to get rid of the external tanks
as well.

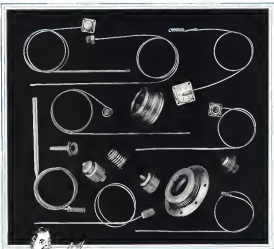
Basic feature of the modification is
the internal mounting of two multiple
rocket launchers. They might fire
independently or be fired singly,
or at a cyclic rate of three per second
from the launchers mounted in the AD-
100 engine.

That means that in 51 sec. the pilot
can throw 1 of a ton of rockets at the
opponent.

But of disposal within one of its own
the rocket motor from the
launcher must follow down and off.

Total weight of the rocket launchers
is 360 lb. Rockets are designated Aero
XRA, are 6000 and distributed in light
in 1000 seconds to the payload.

It would seem that a fair amount of
work would be necessary to make the
Skyrocket under ground under the
launchers and rockets. It would also
seem that this particular installation is
not a precursor for the rest of the AD
line. The next likely step is that
the ship is a single, but modification out of
which will come engineering data for
future use, on what should be clearer
result—DAA.



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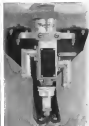
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Special Valve in British G-Suit

(McGraw-Hill World News)

A mechanical device to increase the directed flow of oxygen now has recently been developed on England as an outgrowth of extensive research into the effects of high G loading.

Representatives of the Royal Aircraft Establishment at Farnborough resulted in the design and development of a new G-valve, of which the significant component is a control valve.

► **Compact Unit**—The new valve was designed and built by the Hyman International Co., Ltd. of Rotherham. A compact unit is made of 59 separate, precision-machined parts.

► **Operates from a supply of compressed air at 20 psi**, fed from high-pressure air storage bottles, which may either be charged on the ground (before takeoff) or be charged from the plane's own air compressor unit.

In the valve, a balanced poppet rock arm—controlled by low G acceleration, controls the air supply to the various parts of the unit. A special pressure-relieving device is included to assure that the pressure built up in the unit never becomes so high as to be uncomfortable to the wearer.

► **Flight Tests**—The valve is being tested at present in the de Havilland Vampire-Bristol's latest and most powerful single jet high altitude fighter. At the same time, in order to compare operating characteristics with the two types of aircraft, the valve is also being tried out in the piston engine Spitfire.

Recent problems faced in the development of the unit was the matter of the compressed air supply. This could not be satisfactorily drawn from the compressor of gas-turbine power plants, for that air had to be cooled and its

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pressure reduced and stabilized. For flowover, such a source of supply proved insufficient at high altitudes or when the engine was throttled back before making a dive. In a sharp pull out after the dive, or a tight turn, the fuel would be of little or no use to the pilot. Hence, an independent source was selected for the compressed air-storage bottles.

The valve is designed to be inoperative in shocks of short duration, such as might be caused by hitting an air-seal airfield or by rough air in flight, yet should respond with restoring rigidity to the changing G loads encountered during completed high-speed maneuvers.

NACA Studies Roll Phenomena

Two important added contributions to the understanding of roll phenomena in supersonic aerodynamics have been published recently by the National Advisory Committee for Aeronautics in technical notes.

Laminar theory is used.

• To illustrate the damping in roll of a wing-body combination.

• To estimate the stability derivatives due to roll for sweptback tapered wings.

In both cases, the wings are considered to have supersonic leading edges, that is, the component of velocity normal to the wing leading edge is supersonic.

• Roll Damping—Most previous studies of supersonic damping have been made for a tapered wing. Since such a configuration is rarely found in actual practice, more serious and extensive roll damping for the conventional combination of wing and body had to be found.

NACA's study was based on the consideration of rectangular and triangular wings mounted on cylindrical bodies.

An approximate source distribution was used to represent the interference effect of the body on the wing, the variation reported for this was shown to have small effect.

Two further assumptions were used to limit the analysis. The first is that of vanishingly small wing thickness, which may be expected to have small effect. The second, which NACA considers possibly the more severe of all three limitations, is that of an inverted flat plate due to this assumption is almost impossible to calculate theoretically, and there is not much available experimental data at hand.

• Drag Coefficients—Results of the analysis are presented in charts for rectangular and triangular wings on a cylindrical body. The drag-coefficient derivative is plotted against a chosen velocity ratio of body radius to distance from body centerline to wingtip.

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curves are drawn for several values of aspect ratio.

Additional design information is presented in the form of charts which illustrate the effect of the pressure of the body.

► **Stability Derivatives**—Calculations of the lateral force and yawing moment due to roll (for supersonic wings) are based on the assumptions of supersonic loading, rigid, streamline type, swept leading edges (which may be either subsonic or supersonic provided the latter wing is less than or equal to one).

Further restriction is that Mach lines from either wing must not intersect the other tip.

► **Results**—Final answers are presented in graphs charts for both stability derivatives as functions of a roughback parameter for an aspect and first taper ratios. Both these parameters were varied by the Langley Aeronautical Laboratory Tech Note 2151, "Estimation of the Dragging in Roll of Supersonic Leading-Edge Wing-Body Combinations," by H. Warren A. Tucker and Robert G. Plunk, Tech. Note 2156, "Theoretical Calculation of the Lateral Force and Yawing Moment Due to Rolling at Supersonic Speeds in Sweptback, Tapered Wings with Streamwise Tip-Supersonic Leading Edges," by Sidney M. Hadden and John C. Meier.

Stator Assemblies Made New Way

Development of a simple and economical method for manufacturing compressor stator assemblies for jet engines has been announced recently by Wallwork Patents, Ltd., Lymington, Hants, England.

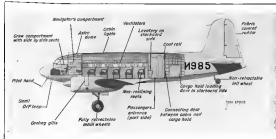
The method, involving a light drawn steel strip section in an aluminum alloy cast by the Al-Pin process. Wallwork is licensed by Al-Pin Division of Fordell Engine and Airplane Corp.

► **Flow Machining**—It is important in the manufacturing of compressor stators that the machining process for the root section. Anything done to simplify the procedure reduces both the time and the cost of machining.

Steel strip used in the process is obtained from the roll in the bright drawn condition. It is then cast by the Al-Pin process in a mold by the roll. The strip is cut to length and bent to the aluminum alloy cast.

Possible future development of a segmented blade unit is foreseen, which a number of blades or even a complete stage may be made in one piece by bending the steel blades to the aluminum alloy root section.

Wallwork has announced that blades manufactured by the process are completely unobstructed. No production forms were mentioned.



SOVIET FEEDERLINER, Yak 16, is being used on feeding Aeroflot system as replacement for older aircraft and other ship to Soviet DC-3 (LJ 2). Designed by Alexander Yakovlev, more famous for his fighter aircraft, the plane is cheap and economical.

Soviet's Newest Feederliner: Yak 16

Aeroflot, Soviet airline, modernizes system with 10 seat replacement.

The Yak 16, Soviet Russia's latest feederline transport, is a low-cost aircraft designed by Alexander Yakovlev—but it is so simple that the Russians have not had time to retract its transport aircraft.

Russians of the Yak 16 is to clear the local traffic, work with the DC-3 (Soviet DC-3) and to replace some of the older aircraft and for short haul.

► **Ten-Place Aircraft**—The Yak 16 is a low-wing monoplane of metal construction with a normal load of ten passengers, three baggage and three crew members.

The plane has a considerable cabin which seats ten passengers on either side of a central aisle. The cabin is not adjustable, a matter of little importance on short hops. There is a large window and an individual window for each passenger.

At the rear of the cabin are the galley and the lavatory and washrooms. Between the rear of the cabin and the cargo hold is a coat rack and a baggage rack.

► **Front Cabin**—The crew compartment is divided into two sections. The pilot and copilot sit in a well-lit flight cockpit under a suspended canopy in a room 8 ft. 6 in. long.

Wingspan is given at 36 ft. length, 36 ft. Gross weight at takeoff is 14,100 lb., of which 10,000 lb. is in the payload.

These performance figures are obtained with two Avia 21 seven-cylinder, inverted, radial engines, equipped with Avia 21 carburetors within their design capacity.



SPARTAN SIMPLICITY of Yak 16 aircraft is emphasized by functional cabin layout, lacking in luxurious dress. Uncluttered appearance of pilot's position is noteworthy.

the blind-flying panel and main flight instruments mounted behind the pilot. All engine instruments are mounted centrally. Captain's panel contains only main gauges.

Control columns are mounted at the outside of each pilot's seat. Brakes are operated by hand from a grip on the port control wheel. Engine, fuel, pump, flap, landing gear and trim controls are positioned mounted between the two pilots.

Both windshields are fitted with anti-ice systems, and there is a direct vision panel for the first pilot.

► **Performance**—Yak 16 cruises at a reported 180 mph at 5,000 ft. Its service ceiling is 16,000 ft. Takeoff and landing speed is 60 mph. Cruise speed is 160 mph.

Wingspan is given at 36 ft. length, 36 ft. Gross weight at takeoff is 14,100 lb., of which 10,000 lb. is in the payload.

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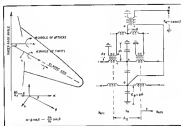
For the same range and at the same speed the Dove carries more payload than any comparable aircraft—and far more economically. It has remarkable servicing accessibility and facility



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AVIONICS



ELECTRICAL ANALOGY for elastic structure of sweptback wing shows equivalent transmission from ducts due to stresses by use of electrical transducers

Analog Computer Aids Plane Design

Caltech electric brain rapidly solves lengthy and complex problems of aircraft structure and aerodynamicity.

Complex problems of aircraft design can be tackled in short order by an electric analog computer operated as part of the Analytical Laboratory at the California Institute of Technology. Similar, but on the many-branched family tree of calculating machines, the analog computer is a simple, self-powered, economical device. It will solve problems not demanding of better than one percent accuracy, and solve them rapidly.

Caltech's computer was the first of its kind to go into service, now, it has three years experience in solving the aircraft industry.

► **Principles**—Fundamental operating principle of the electric analog computer is that analogues exist between the components in mechanical systems and electrical circuits.

Take for example a mass inertia (mechanical) and an inductor (electrical). The mass tends to maintain a constant velocity because of its inertia, the self-inductance of a coil tends to oppose any change in current. By comparing resistance for the inertia force and the self-inductance voltage, analogues between force and voltage, and velocity and current can be shown to exist.

► **Classical**—By using the complete set of analogous electrical components, a mechanical system can be set up as an electrical circuit with properties exactly analogous to the mechanical properties of the system being studied. These circuits have been developed to the point where they are suitable for a very wide range of applications, including such dynamic cases as transient heat flow, in

transient heat flow problems. The all models very easy—and it is, the simple system. But the advantage of the analog computer is not that it solves simple problems, but that it will quickly and easily give the answers to those very complex ones. And for those the credits are not so simple. They require some time to work out.

► **Disadvantages**—In Caltech's computer, electrical components are available through the medium of plugboards for circuit element selection. Once the elements have been determined, they can be set up quickly with these boards.

In contrast to other types of electric analog computers, electrical resistors, inductors, capacitors and transformer circuits are used by Caltech for simulation of the basic laws of algebraic and differential equations.

Known functions of the independent variables are impressed on the system by forcing functions, in the form of steady state, sinusoidal, variable-frequency, square wave, transient or even completely arbitrary functions of time. Amplifiers represent negative impedances. Non-linear expansion of the equations are handled by the multipliers which can multiply any two variables. Eleven arbitrary function elements form dependent variable functions. Any special non-linearities (inside reliable limit signs, for example) are available through five constant or self-adjusters.

► **Accuracy**—Solution of the problem is obtained by measuring the output voltage, current or charge as a particular output.

In the case of transient solutions, the answers are displayed or recorded by cathode-ray oscilloscopes. Steady-state solutions are measured either with vacuum-tube or displacement-type meters.

Now, of course, the aircraft engineer is interested in what sort of problems can be handled by such a computer. The first answer is largely self-evident—those which take a great deal of time being done. But there is a modification to the first answer which is also important, and that is that the problems should not demand a solution accuracy of better than one percent.

This is not really a limitation, however, because the general problems of aircraft design rely on data and assumptions that, on a good day, are possibly correct to within five percent. And in that category come such areas as vibration and aerodynamic problems. Analysis of automatic control systems would be another genus of problems suitable for such a computer.

► **Complete Analogue**—The elastic structure of an entire aircraft can be set up in detail and analyzed. Wings, fuselages and tailplanes can be represented either as beams with combined bending and torsion, or as vibrating structures plates. Beam equations are solved in finite difference form; then solutions exist for bending deflection, slopes, angles of twist, shear, torque and torque loadings.

The structure thus simulated may be analyzed not only for static loading conditions, but even for the very complex transient conditions obtained, for example, during flight through gusts. Other transient analyses would include landing and braking shocks.

► **Three-Year Service**—Full time use of the Caltech computer has been going on for about three years, during which time a great number of aircraft problems have been solved. Enthusiasm among those with a complete airplane



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vibration analysis, wave vibration analysis and gust loading for Douglas Boeing and Lockheed analysis for Lockheed, and tunnel vibration study for Caltech and a rocket vibration problem for NASA.

Capacity of the computer has been doubled since operations began, and it is possible to add more. Doubtless the use of the computer has been so many that there has not been enough time available, even with the doubled capacity, to handle all the requests.

Knowledgeable—Thomas Wynn is indebted to Dr. G. D. McCross, Director of the Analytic Laboratory at the California Institute of Technology, for background material furnished for this article, and is responsible for certain wave and data taken from Dr. McCross's data.

Dynamic Stability, Control Simulator

A dynamic stability and control simulator has been developed for the Navy by MIT to perform the dynamic calculations involved in analysis of aircraft flight characteristics. The new machine is expected to reduce the time, expense and number of conventional flight tests by determining flight characteristics in advance.

The flight simulator consists of a battery of computing machines and a "flight table" consisting of an arrangement of gimbals suspended so that they can incline freely in any direction and supported on an independent foundation to eliminate vibration. The gimbal frame is operated by very high speed hydraulic servo mechanisms which automatically control instruments that carry out various in simulation with electronically transmitted commands. It is used to erect the automatic control system of a theoretical plane or missile just as it would be tested in actual flight. The motion of the gimbal frame table can be recorded for study by engineers.

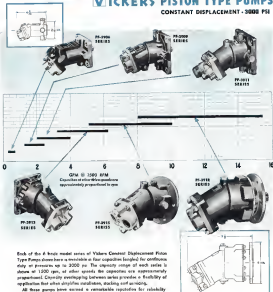
A problem is set up on the computer by setting electronic computer data that represent such characteristics as weight, speed, altitude, wing area, etc. Then the computer is fed into the simulators, by applying appropriate electrical signals through a control board. The answer is returned on a chart as a recording apparatus in a matter of seconds.

The project was headed by Dr. Albert C. Hall, director of the MIT Dynamic Analysis and Control Laboratory. Among MIT scientists contributing to the project were Dr. John F. Black, development of automatic controls and supervision of mechanical design of the gimbal frame; Tracy S. Gerner, Jr., development of instruments

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tion of men and goods—no more fully understood. So are the difficulties of obtaining it. The problem has been pushed by a Broad-based Commission and a Congressional Board. And the major findings of these bodies have been generally accepted by the Congress. As a matter of fact, the present congressional aviation program you guess may well be considered a belated attempt of the government to catch up with the recommendations of these bodies, with the action of Congress and with public opinion.

► **Planned Expansion**—Major changes have occurred since 1940 in the airplane and propeller industry, major ones in the engine industry. Among the leading airplane manufacturers in lead were McDonnell, the Douglas, and Boeing. Some companies, such as Vega, Vultee, and Stearman, have become part of larger companies. But on the whole the plans of tomorrow will bear World War II expansion names.

The jet engine, however, has brought two new and major products into the engine field. General Electric and Westinghouse. The fact that jet engines have never been mass-produced is the reason that expanding engines were during World War II puts a great question mark behind all other things below. For that matter, jet engines do pervade down the line in airplanes, engines, propellers, aviation equipment as aviation may throw the paper oil columns out of the window.

► **Flow Space**. Total floor space of airplane manufacturers in 1959 was only 7.5 million square feet, but in 1940 it had grown to 9.5 million at the beginning of 1944 it was around 110.5 million today it is probably 41 million, and growing the steadily plants.

If we compare the floor space of an average prime contractor in 1940, 1944 and today we find a ratio of 1:11.5 in 1940 and 1:4.7 today.

► **Employment**. Employment by an average prime contractor shrank from 68,700 in 1940 to 50,100 in 1944. In 1940 it was probably as high as 100,000. This gives a ratio of 1:6.3. Such a ratio, however, is badly distorted because of the strong effect of rationing during in 1944.

A rough estimate, taking into account employment by engine case makers, subcontractors and parts suppliers of airplanes, engines, propellers, gliders, and special purpose aircraft for June 1940, June 1944, and June 1950 produces a ratio 1:15.2.

*All data and ratios are based on data submitted by the government to AIA. Data for the engine industry, data for "other contractors" should be adjusted for sub-contractors. While submitted under 40 percent in 1940 and today, manufacturing data in 1940 was in excess of 20 percent.



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A.B.C. REPORTS — FACTS AS THE BASIC MEASURE OF ADVERTISING VALUE

• **Production.** Production of military aircraft presents a disappointing picture. In November, the rate for 1944, 1944, and 1950 will probably be about 1158.0.5, representing 6180 planes produced in 1946, 56,559 in 1944 and about 3000 in 1950.

More significant—and somewhat less discouraging—is the ratio for average weight (including spares) produced in the same year. Roughly, the ratio may be estimated on the basis of a 1940 production of 24.6 million pounds, a 1944 production of 1.1 billion pounds and an estimated 1950 production of more than 48 million pounds at 1.45 : 1.

The wide difference between the number and the weight may be explained by the rapid increase in the weight of the average aircraft. This is due to the shift from the production of the lighter fighters in 1940 towards the production of the heavier fighters and bombers in later years. It is also due to the constant increase in the average weight of each type of aircraft model.

• **Payroll per Worker.** Trends of military aircraft assigned per employee, adjusted for subcontracting and including the weight of spares, grew from 21 pounds in January, 1941, to 24.6 in June, 1941, to 26.5 in March, 1942. The 1944 average was 88.2 pounds per month. Today the output per man per month is probably below the 1941 figure. An estimated rate of 1.45 : 1, for 1940, 1944 and 1950, would probably not be too wrong.

• **Planes & Missiles.** The immediate availability of greatly increased facilities, both in size and in number, appears to be the greatest advantage the industry has today over 1942. This is the direct outcome of wartime and postwar expenditures of about \$4 billion by the government and by the industry which were directed at construction and equipment. These facilities, together with a reserve of machine tools, should allow a much rapid increase in production than was possible in World War II.

There are also more—and better trained—employees, plus a considerable number of workers with some wartime experience in aircraft plants.

On the negative side must be listed the fact that the industry may be called upon to accelerate its production at a time when manpower and materials may be seriously difficult to obtain on a large scale, and at a time when the shift and the call of the reserve actually threaten its available manpower.

Another negative fact is that production has been allowed to drop to extremely low levels from which recovery is only now coming.

• **Government Proposals.** The institutional confusion, conflicting as-

serts, lack of program, annual scheduling and material mismanagement of the early days of the war are not forgotten while we await catastrophe. Only practical experience will be able to tell if the government is better prepared today to cope with a new emergency more quickly.

The existence of the National Security Resources Board (and the fact that it finally has an aggressive head), of the Munitions Board, and of the Industrial Mobilization Planning bodies of the military departments give hope that this is the case. So does the fact that industrial mobilization contracts have provided some made plans for the mobilization of major prime contractors.

On the negative side must be listed the hesitancy in being necessary controls on the top levels of the government and the hesitancy lack of early scheduling of military aircraft procurement once the war. The latter seems to indicate a lack of comprehension or of application of reality between the mobilization studies should have disclosed.

• **Industry Preparedness.** Obviously, the aircraft industry is much better prepared today to cope with an emergency than it was in 1940. At that time the old Association of Aircraft Manufacturers was limited in its representation of the industry and in its resources. In addition, the highly competitive situation of the industry made it unlikely that any competitive venture would have much chance of success in the early days of the European war.

Actually, it was the Tuscan Committee, which, after being President Roosevelt to appoint an aircraft industry, hastened the creation of the West Coast Aircraft War Production Council in April, 1942. Pioneering cooperative practices for the defense of the war, various tasks available in construction, to the government, and to contractors all advancing which helped war production. The West Coast Council was followed by one on the East Coast and later by a national one.

In 1944, some of the functions of the Council's were transferred to the present Aircraft Industries Assn., and the Council ceased with the end of the war. The present representatives of AIA would need only little expansion to assume the functions which AWPIC carried out during the war.

In case of an emergency which would require all-out production once more, it will become a question of policy for the emergency presidents to decide if these activities should be carried out by the permanent trade association of the industry or should be transferred to an emergency body which could assume the quasi-governmental status which the AWPIC maintained during the war.



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LETTERS

Milt Arnold on ILS

As article by Capt. B. C. Wilson appears in the June 12 issue of *Aircraft Week* entitled, "Need Improved ILS Monitoring" indicated serious deficiencies in the ILS approach system. The Air Navigation in Traffic Control division of the Air Transport Act has actively followed the same during progress. In view of the statements made in the article, the operational difficulties to close up some necessary equipment which may have resulted from making the article.

The article states that the need for two factors to make the reliability of the ILS, the fact being fully automatic monitoring and the manual monitoring of a monitor at a point in the approach which will check the location along the entire approach course.

With regard to the first point, the CAN has under way a program designed to provide fully automatic monitoring of the location component. This program has not with considerable delay become the construction facility to produce satisfactory equipment. All pilot duty transmitters are monitored automatically at the present time.

The reference made in the article to an error criterion which occurred during ILS approach, and the reference that these errors were caused by failure of the monitoring system to indicate ILS malfunctions, seem to me to be entirely unexplained. Due to the need for a reliable statement on the error rate of the overall monitoring location of flight because they were approaching with lowered gear. There has been no evidence presented in any of these cases which specifically indicates ILS ground equipment malfunctioning as monitoring basis.

The suggestion made in the article to monitor the location at a point along the course is by no means a new idea. During the early stages of the development of the ILS, this idea was considered, and the system and tests were made to study its feasibility. With the introduction of the ILS into the European Theater of Operations in 1944, the idea was considered again, and tests were made in an attempt to obtain satisfactory results of monitoring the local ILS. The operational results of these tests were the results of these attempts to monitor location at the approach end of the runway have consistently resulted in failure.

It is true that the ILS can be replaced very simply. It is, of course, not feasible to place a monitor antenna at an elevation along the approach path which would be that of an antenna on an aircraft being on ILS approach. Practical considerations state that the antenna must be well below the full wave height of the glide path in order not to present an obstruction.

These findings with the propagation characteristics of the VHF frequencies used in the ILS system are aware of the high degree of attenuation which occurs when a jockey antenna is lowered close to the

ground. A monitor antenna, for instance, located at the middle of the air deviation of ILS would receive a small fraction of the energy received by an aircraft antenna at an altitude of 200 feet above the ground. This low field strength results in a high degree of susceptibility to reflected signals. The ratio of reflected signals to direct signals received by the antenna antenna at the 15 foot elevation is very high and very accurate or obstructive in the way will cause signal changes and variations in the course of the practical program, it is impossible to obtain reliable information from a monitor antenna at this location. Furthermore, the variations encountered here have little relation to the variations encountered in the entire approach of the course.

The *Aircraft Week* article relating to sweeping corners encountered at La-Moche based on marine recordings taken at the night station cannot be regarded as having any real significance. The present standard for location which is about 110 feet from the transmitters indicates that the result of any error in following the most suitable location for accurately determining the air delay at the time.

The heads which are in location systems are well understood and also used previously from large observations such as terrain and power lines. They are most pronounced along the observation, as located to cause proximity to the terrain. The closer the observation, the larger the period of time. Practically all pilots are aware of the limitations of the observation during a location approach when actual altitude takes off directly over the location transmitters. This indication shows that a period of approximately one to two seconds and although it is violent, it is well understood that it is not considered hazardous by most pilots.

We also disagree with the criticism of the use of GCA as a monitor for the ILS. The two systems are by nature mutually complementary and when used together, they provide the safety of each system. The fact that in certain cases the ILS path has not agreed with the GCA path is a deficiency of a particular installation rather than of the concept involved. The use of GCA as a monitor has proved its worth on a number of occasions and it, in no opinion, is not a necessary addition to the overall landing system.

We would like also to correct an expression given by the writer on the subject of propagation in radio frequencies. Serious attenuation due to path length effects does not begin to come into frequencies of over 3000 mc/s or more.

It is not a particularly important factor one of these frequencies except in the case of primary radar where the amount of reflected energy picked up by the receiver is extremely small. In fact, the attenuation occurs in the space between the reflecting antenna and the aircraft receiving the signal, and as ILS is concerned, the speed of the aircraft has to be taken into the amount of attenuation encountered.

It is likely that this expression may have resulted from faulty antenna fittings or some portion of the aircraft receiver input circuit which was affected by noise.

Although it is recognized that some operations and static is encountered with the frequency used in ILS, the degree of the propagation delay at VHF is extremely small compared to that of an HF and LF propagation delay in a very significant way. It is a fact that a few seconds are VHF while at lower frequencies radio waves are slower with serious interference from this source. For all practical purposes, propagation delay at VHF can be neglected.

It is no system that the present position of the CAN for the improvement of location system is generally sound and that much of the criticism contained in the *Aircraft Week* article was not justified.

MILTON W. ARNOLD,
Vice President—Operations
Air Engineering
Air Transport Association of America
Washington, D. C.

(Capt. B. C. Wilson, author of the article discussed here, is an American Airlines pilot—E.A.)

Free Enterprise?

First, I will introduce myself. I'm Larry Atkinson, editor and publisher of *Wings* Air Lines.

We wouldn't call it a free press, but we do have editorial independence on the "Air Column." You may have noticed that we've been publishing the views of the public which the general public should have the opportunity to read.

Now editorial Mr. E. J. Wilson, "Chicken Man" for GEC, says "The readers were among a few million." That is absolutely correct. It is now the matter of the circle through you and I have placed a few of the valuable items in the "Wings Market." Let's all give a big cheer to these able executives who can and will arrange from the bottom up and give us the best of times when we can all enjoy the good available.

But what about the position—the standards that feed the way the public to stay as much?

Must they be thoughtful—put out or edited? What's a lot for editors? I refuse to be a victim and I don't order much by GEC to Wing Air Lines.

We'll assume your place for a second and legal advice.

A. J. BUCKENHAM, Secretary
Wing Air Lines
Burlington, Calif.

Praise

We look forward to the receipt of *Aircraft Week* each week and depend almost entirely upon its contents to keep us in touch with the progress in the aircraft industry.

Conrad D. Dwyer, President
American Helicopter Co. Inc.
Meriden, Conn., Calif.

On the go? This Beechcraft lets you accomplish more!



Speed and Top Performance

Because you cruise at 170 mph, all the travel time you formerly wasted is put to rest in a new way. You'll enjoy a top speed of 170 mph. The Bonanza's 170-mile speed gives you a lot of freedom.



Extra Speed When Needed

Want more speed? No problem. You can cruise at 170 mph. The Bonanza's 170-mile speed gives you a lot of freedom.

With automatic power, you can cruise at 170 mph. The Bonanza's 170-mile speed gives you a lot of freedom.

A. J. BUCKENHAM, Secretary
Wing Air Lines
Burlington, Calif.

There's much more to tell about this versatile, economical business plane.

For the full story on the revolutionary Model B20 Bonanza Bonanza, contact your nearest Beechcraft distributor or dealer. Or for more details, write Beechcraft Aircraft Corporation, Wichita, Kansas, U.S.A., or your company letterhead today.

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Short-Field Performance

Because of their shorter landing gear, you can land on a 100 ft. runway. You'll enjoy a top speed of 170 mph. The Bonanza's 170-mile speed gives you a lot of freedom.



"Passenger Car" Economy

Save the Bonanza over 30% of the operating cost of other business planes. You'll enjoy a top speed of 170 mph. The Bonanza's 170-mile speed gives you a lot of freedom.

With automatic power, you can cruise at 170 mph. The Bonanza's 170-mile speed gives you a lot of freedom.

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Wing Air Lines
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Stable in-Flight Control

Exceedingly superior control over the air with a "steering wheel." You'll enjoy a top speed of 170 mph. The Bonanza's 170-mile speed gives you a lot of freedom.



It's Strong on Safety

Sturdy, low-maintenance landing gear with no need for "steering wheel." You'll enjoy a top speed of 170 mph. The Bonanza's 170-mile speed gives you a lot of freedom.

With automatic power, you can cruise at 170 mph. The Bonanza's 170-mile speed gives you a lot of freedom.

A. J. BUCKENHAM, Secretary
Wing Air Lines
Burlington, Calif.

Top speed, 174 mph

Cruising speed, 170 mph

Range, 750 miles

Fuel economy, 9.5 gal.

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New low cost aircraft spark plug assembly designed by Auto-Lite for light plane use. Has a high-quality 16mm Aircraft Spark Plug with automatic type shielded-shell to give long life and low cost service.

16MM SHORT REACH SHIELDED

The 16mm Aircraft Spark Plug was especially designed by Auto-Lite to give long life service under the most severe operating conditions. Suitable for higher power engines requiring a superior type of shielded spark plug up to 1,000 H.P. Military aircraft buy a better shielded spark plug.



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Auto-Lite developed the 16A-1 for the most rigid requirements of aviation aircraft. This precision-made 16mm plug has built-in type metal, multiple electrodes, one-piece shell design, corrosion resistance.

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The new Auto-Lite Resistor Spark Plug reduces radio interference and radar interference normally caused by ordinary spark plugs. Get smoother performance — money-saving too — economy — less power starving in cold weather. Install this essential plug on all your ground equipment.

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Auto-Lite Spark Plugs — Federal (U.S.A.)



Approved for use in aircraft by the Federal Aviation Administration. This plug is built to meet the most rigid requirements of aviation aircraft. It is designed without preliminary single engine test.



SALES & SERVICE

Services Eye Aero Commander

Twin-engine executive plane versatile, maker says; would make good trainer; could take jet engines.

A versatile twin-engine prototype plane with a new CAA certification is making an all-out sales demonstration for potential Air Force and Army buyers at Washington National Airport.

Developed as a 5-7 passenger executive transport at 4,100 lb. gross weight, the Aero Commander is now being demonstrated to the military community as a "staff car" plane with twin engine safety.

Among other uses for the plane are light cargo carrier, air evacuation and, as rescue, supply drop, low flying, staff transport, gunnery or bombing trainer and target towing.

Specifications for the all-metal Aero Commander has been generally described previously in *Aircraft Week* (July 11, 1949 and Nov. 25, 1949). It meets essential specifications and details are:

Alternate powerplants: Licensing 190 hp. 9-45-1 engine, now fitted; Lycoming GO-155-C engine (general) rated at 260 hp. for inland, Continental E-225 engine with associated power at 225 hp.; or the jet and turbo-prop engines previously discussed.

Selected guaranteed performance data (with 190 hp. engine): maximum air speed 260 mph, cruising speed at 75 percent power, 165 mph, maximum climb speed, 65 mph, stall speed flap and port down, 56 mph, maximum at cruising speed with 165 hp. fuel, 5,475 hr. time at climb, first crew four sea level, standard atmosphere, 14,470 ft. cruise, two-engine service

ceiling, 22,190 ft.; stable ceiling, single engine with windmill propeller, 5200 ft.; landing gear 50 ft. obstacle, 1113 ft.; landing roll to stop after clearing 30 ft. obstacle, 1300 ft.; cruising speed at 18,000 ft., 170 mph.

While only one engine has been tested, the engineering company now has substantial success in building and is prepared to produce additional military and civilian models at Oklahoma City, Okla. It already has some components of additional planes and some testing made for use. Arrangements have been made to sub-contract construction of some components by Lomax Aircraft Corp. and other manufacturers. Deliveries of production planes could be started by mid 1951. Southwestern Aircraft Co. is building a 51A-100 to 51A-1000 has been put on the airplane, depending on the specific personnel, equipment installed and the particular model.

Engineering development of the plane was started by Smith and has West Coast engineering associates in 1947, but actual production did not begin until Aug. 1948, and the prototype made its first flight April 25, 1948.

The plane is certified to meet stability, maneuverability, and control requirements of Civil Air Regulations 13 in all flight conditions. More features state that from the date of the first flight until certification was completed June 18, 1949 not a single basic change was made in the engine.

Several Washington demonstrations to Army, Air Force, Navy, CAA, CAA, Department of Agriculture, Forest Service, and other agencies, plus plans made to Fort Monmouth and Fort Rucker for Army demonstrations and to Portsmouth for a Navy demonstration. Following the Washington demonstration it is expected to be flown in the Standard Commercial plane at Wright Field.

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DEPOSITOR WITH WINGS

Four Indianapolis postmen serve as a Ryan Navyfin set up a bank account in the name of this plane to be used in paying operating expenses. Each postman pays into the fund for living base, with overhead costs de-

mined in proportion to total hours each owns the plane. Each expense is paid in dollars equally. The check shows above it in old form, postmen also have the number of these 1950 plane, 5111K.

Mobilization's Effect on Airlines

Investors feel that all-out effort would mean peak domestic load factors and high plane utilization.

The expectation that any full-scale mobilization of the national economy will follow the same pattern as World War II is increasing investor attention as the airlines.

The air transport industry did very well in World War II from an earnings standpoint, despite (and because of) its activities which were in those years.

The scheduled airlines are currently only partially committed, with 45 of their planes in charter operation for the military in the Pacific zone.

It is known that the air transport industry has evolved plans to supply additional aircraft to the military, on contract, should emergency needs demand it.

Final decision on the disposition of the commercial air fleet during an all-out emergency rests with the National Security Resources Board, which is charged with coordinating and obtaining the maximum utilization of military, industrial and civilian facilities and manpower for war.

The National Security Resources Board has not yet ruled on this dilemma. Presumably this decision will be reached after existing legislation is sought to implement mobilization plans.

Military Role. It is hoped to ensure that, during wartime, most U. S. transportation modes will be operated directly by the military or by the airlines under contract to the military. On the domestic front, the commercial carriers hope to preserve their identities as they deal during the last war.

The air carriers now fully expect the diversion of most of their four-engine aircraft to military operations as mobilization plans take effect. The full degree of such diversion will be influenced largely by the degree of emergency.

In addition, there are about 670 two-engine aircraft available among the scheduled airlines for domestic and international commercial use. Most of these planes instead of having as many as four engines consist of 21, one-engine planes for less than 25 ft. Greater speeds and fewer hours would also increase their availability.

Long-Haul Needs. The commercial airlines will seek to keep as many of their four-engine aircraft as they can for long-haul domestic routes. There are potential operational problems if the airlines

have only two-engine equipment at their disposal.

For example, a DC-6 in Constellation or an increased long-haul with a 3-man crew is capable of producing 11,000 ton miles per day per man (averaging a 10-hour utilization), or delivering 15.2 ton miles per gallon of fuel consumed. A two-engine DC-3, under the same assumed conditions, could produce but 2100 ton miles per day (with a 2-man crew) and average only 4.7 ton miles per gallon of fuel.

These comparative rates merely highlight the lower direct or manpower and fuel two-engine commercial aircraft could cost. These inequities are prohibitive in time of national emergency and would almost automatically result in the airlines' rejection.

Proponents of the commercial airline viewpoint assert that during wartime most of their traffic will be high priority passengers, cargo and mail. Moreover, it is essential to keep open the key lines of transportation and communication provided by the airlines if industrial production is to proceed at a maximum rate.

Flare-ups in Personnel. Another concern during a period of national emergency would be the allocation of the number and type of aircraft each carrier was permitted to operate.

A rise in future revenues under two utilization conditions can be achieved from past experiences. Of about 515 transport planes in domestic service as Dec. 7, 1941, almost half were registered by the military. Nevertheless, with but 51 percent of their former number of aircraft, the airlines were operating 71 percent of their former mileage.

Despite the limitations imposed by the experimental shortages, the airlines continued to show increases in every revenue element during the four-year war period.

A combination of circumstances was responsible for this amazing showing in the first place, there was no problem of flight plans to capacity with airlines willing to put up with almost any sacrifice.

The revenue factor came into active play on the air-side under such conditions, and translated all revenues above the break-even point to profit during World War II.

Four load factors were considered satisfactory when operating in the 70s. With wartime traffic being maintained at capacity levels, load factors hit 90 percent and more. The impact on earnings was obvious.

Higher Utilization. Greater utilization was obtained from each plane. For instance, instead of flying less than seven hours daily, planes averaged closer to ten hours during the war years. This permitted the carrier to fly an average of 1601 miles per plane per day during 1944, for example, with an average utilization of 18.49 hours daily.

With depictions and other fixed expenses remaining, whether planes flew or not, the importance of this higher utilization is quite evident. The peak high utilization of airlines equipment during the war years has remained unsurpassed thus far during the postwar years.

For example, during 1948, the industry's daily average utilization averaged 6.41 hours, resulting in an estimated 1990 miles per plane per day—the lowest utilization since 1943.

Unknown Factors. There are a number of unknowns, however, which can alter the airline's optimistic outlook under full mobilization conditions. The industry will be subject to the same normal tax impacts as apply to other businesses.

Blank assumptions must exist that the airlines will receive the same favorable treatment under new excess profits tax law as during the last war. At that time, the Internal Revenue Code provided, in substance, that as long as an airline's adjusted net income did not exceed its gross mail pay, it was exempt from excess profits tax. (Factors was the only domestic airline which paid this tax during the war.)

If the same provisions were to be repeated in subsequent economic periods, the airlines, as a group, could earn more than three times their 1949 net profits before being subject to the special tax burden.

The ultimate degree of satisfactory prospects on airline operations must be given greater weight at this time than in the past.

The present threat of invasion in the form of wages and capital can be cause for more diagnosis of the country's economic resources involved in an inflationary spiral. Unless the industry can adjust its relationship fixed rate structure to absorb any potential price increases, it will be in a tight spot in the event of changing circumstances.

The airlines, because of their peculiar leverage and attendant leverage, do not always follow the course of past patterns even when many things are the most obvious. Each phase of their development must be appraised in the light of changing circumstances.

—Sally Altshuler



1908 ★ THE MIRACLE OF AMERICA ★ 1950

It's no stretch of the imagination, rather, robust realism to call our past half century a Miracle—U. S. A.

America has set an amazing record of progress in 50 years—but a moment in the history of civilization. A record unequalled by any other political or economic system.

Merely by broad brush strokes, we can all visualize this miracle. Remember the crystal set, the hand-cranked car, the biplane? A far cry from our FM radio, television, hydro-matic drive and supersonic planes.

And here's another phase of the miracle that went hand-in-hand with these and the myriad of intertwined technological advances—ranging from the radio telephone and Bakelite to the X-ray tube and teletype . . . and to atomic energy and its untold potentialities.

- ★ Since 1900 we have increased our supply of machine power 45 times
- ★ Since 1900 we have more than doubled the output each of us produces for every hour we work.
- ★ Since 1900 we have increased our annual income from less than \$3400 per household to about \$4000 (in dollars of the same purchasing power), yet . . .
- ★ Since 1900 we have cut 18 hours from our average work week—equivalent to two greater average workdays.

How did we do it? The basic cause for this composite miracle has been the release of human energy through FREEDOM, COMPETITION and OPPORTUNITY. And one of the most important results is the fact that more people are able to enjoy the products of this free energy than in any other system the world has ever known.

THIS IS THE MIRACLE OF AMERICA . . . it's only beginning to unfold.

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DC-6A illustrated Douglas ideas on commercial cargo plane design: Proven performance, high speed, very large lift door.

Faster Planes Can Lower Cargo Rates

Douglas study shows how direct ton-mile cost is key to airfreight economy and suggests turboprop power.

Some well-accepted beliefs on one-rear cargo plane design have been tested by an engineer in a position to prove his words.

Warren T. Dickinson, assistant to the chief engineer of Douglas Aircraft Co.—the sole manufacturer now flying a plane designed expressly for conversion into cargo—has carefully analyzed and measured such doubts as two statements that have recently become axioms among air cargo experts:

■ **Greater speed should be about 300 mph because cargo rates could "justify the cost of such greater speed."** Not so, Dickinson argues. Greater speed to cut the direct base is important for a large overhauled transport—but cuts also in the direct ton-mile cost.

■ **Cargo plane doors should be at truck-bed height to save time and cost in loading and unloading.** Not necessarily so, Dickinson says in effect. Truck-bed loading may save some time, but without it, an airplane can be more efficient and therefore more economical.

■ **Turboprop for Cargo.**In a paper prepared for the California Air Freight Clinic scheduled last week at Berkeley, Calif., Dickinson, for what may be the first time, proposes a turboprop-powered cargo plane to get the greater speed.

Douglas has studied that proposal,

he says, and concludes that a turboprop cargo plane will be at least 10 mph faster—and reduce the direct ton-mile flying cost approximately 25 percent below those of presently available types.

That calculation, of course, assumes continuing progress in turboprop development, and Dickinson adds that it is not possible to say when a turboprop cargo plane can be built. But he doesn't doubt the possibility of such a plane.

"The high power and low engine weight of the turboprop power plant permits increasing not only the gross weight," he says, "but also the payload and the block speed, while reducing the cost per hour slightly due to the increased fuel consumption. The net result, however, is a substantial reduction in the cost per ton mile."

■ **Important Costs.**Arguments about a cargo plane's speed have generally been based on the importance of the time in transit to the shipper. Considering the time of enroute and surface transit, port time, and distance to be covered and the cost of high speed, cargo aircraft for some time has adopted 500 mph as a good working figure for desirable cruise speed of a cargo plane. This was the speed recommended by the Civil Transport Aircraft Evaluation

and Development Board, a government group studying transport plane needs.

Even a slow cargo plane profits on the surface transit time, as Dickinson agrees that along a few hours more of the transit time isn't the biggest advantage the shipper gets from fast cargo planes. The most important benefit—for both shipper and operator—is lower ton-mile direct operating costs. This also can be proved by another three queries, D. A. Beck, of Boeing.

In Dickinson points out, airlines and government seldom all direct costs on a single basis. He says "an economic basis of the actual flying costs of five representative airlines shows that approximately 75 percent of the direct flying costs are truly 'ton mile' costs. So he set up this formula:

Direct cost per ton mile =
Direct operating cost per hour
Block speed in mph

As demonstrated by that equation, Dickinson says, "The greater the lowest possible cost per ton mile will be achieved by the airplane having the highest block speeds and lowest net resistance. For this reason, aircraft manufacturers are inclined and will pay large dividends by manufacturing and operating."

■ **Need for Efficiency.**"It is obvious that the future commercial cargo plane cannot exhibit any truck-like characteristics in the air and still provide the lowest cost transportation. The revenue importance of economy in the cargo expense demands the highest pos-



LARGE DOOR of DC-6A and hydraulic lift (left) loaded; and unobstructed hold will take cargo 72 ft. long (right).

sible structural, aerodynamic and operational efficiency.

It is just for such reasons of efficiency that Dickinson objects to the truck-bed height philosophy. "The average truck-bed height," says Dickinson, "is about 45 in. from the ground. To drop the loading to low requires a high-wing design for any airplane carrying more than 35,000 lb. (because of propeller clearance)."

A high-wing design immediately establishes two more requirements: a longer, and therefore heavier, landing gear, a heavier landing structure to take loads supported by the wing in a low-wing design.

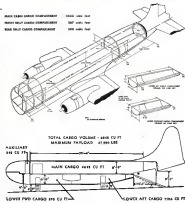
That is no theoretical calculation. According to Dickinson, Douglas designed a high-wing version of the DC-6A. Such a plane would have had an empty weight increase of more than 3000 lb. due entirely to the high-wing design.

■ **What Is Loading Cost?**—But it might be possible that time in air in loading and unloading a high-wing, low-loading plane might overcome the weight difference. So Dickinson examines that plane, too. He says "Several sources were questioned on the savings in loading costs possible on account of the reduced lift, and it was agreed that a figure of 10 percent appeared reasonable."

Based on the actual average haul of 1415 miles (two air miles in May, 1950) and an average loading cost of \$1.25 per ton, the saving will then amount to about \$480 per month for an air freight operator doing 2 million ton miles of business a month.

Since \$480 is only approximately two-thirds of one percent of the direct flying costs for 2 million ton miles of work, it is clear that the best flying craft are much more important than the loading work."

Starting with a basic, proven airplane,



TWO DESIGN philosophies on cargo planes are apparent in these drawings of the Douglas DC-6A (top), with side-side loading, and Boeing Strato-Lifter (below), with various openings.

with which most operators are familiar—the DC-6A—Dickinson designed a new loading to heighten loading requirements. It achieves all loading conditions are adequately met by the DC-6A's very large (176x76 in.) rear cargo door and a forward door 91x67 in. Another method of adapting an existing passenger plane to cargo needs was explained

to the Clinic by Beck, commercial representative of Boeing.

■ **Boeing Approach.**The Boeing Strato-Lifter can be used for commercial cargo, but the only plane of that type now flying size military C-95. Considerably larger than the DC-6A (with a total cargo volume of 6548 cu ft., as opposed to the DC-6A's 1900 cu ft.),

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competition traffic—passengers on U.S. jet carriers claim to prefer service from both TWA and PAA—will go up from 1945's 51 percent to 69 percent.

■ **Foreign Airlift**—One of the outstanding reasons all along for Pan Am's desire to join ACA is that the increasing competition from foreign airlines drove PAA left that with one loss carrier the other two would be in a stronger position to combat the foreign competition.

PAA and figures of the International Air Transport Association to show the effect of the merger on the foreign-air competition situation. In 1946, foreign lines carried 167 percent of all trans-Atlantic traffic (both ways). By the first six months of this year, that figure had risen to 151.6.

Meanwhile, ACA's share of total Atlantic traffic had declined from 27.7 percent in 1945 to 29.1 percent in the first half of this year. PAA's share from 32.0 to 32.5, and TWA's share from 25.0 to 22.8.

Pan American Airlines says that, assuming it and TWA now split 50-50 the London-Panama and Rome-Rome traffic, the future should show PAA getting 51 percent of total trans-Atlantic traffic, TWA getting 31 percent, and the foreign lines still rising to 36 percent.

Hearing Discusses Prototype Proposals

Competent recommendations gathered last week behind proposals for a full scale government-sponsored prototype program to speed development of all-weather transport types and build up national airway capacity.

■ **Three Senators** (McClellan, Ladd, Pepper) introduced an amendment to the proposed \$12.5 million program under which Civil Aeronautics Administration would finance the testing cost of new cargo and transport planes. They said they would push for a larger program.

At last they threatened to block

the testing program, but later withdrew their opposition, and it was passed by the Senate.

■ **Rep. Lindsay Bicknell**, chairman of the Transportation subcommittee of the House Interstate and Foreign Commerce Committee, said: "Antagonism exists that he would issue on 'a program which will assure that our requirements in commercial aircraft are met—both as to types and numbers.' Bicknell, the sponsor of a bill authorizing the expenditure of \$50 million annually for aircraft research and development by the Navy, Air Force or CAA, has already stated his dissatisfaction with the testing program (House News, Aug. 14), now pending before his subcommittee.

■ **John DeWitt Ramsey**, president of Aircraft Industries Association, emphasized on the Civil Coast with numerous recommendations on an expanded prototype program.

Bicknell's subcommittee requested AIA to submit a proposal which would guarantee that the U.S. gets its last position in commercial aviation. Ramsey testified that Civil Aviation was a directorate task to get and develop types.

■ **Secretary for Air Thomas Fiedler** emphasized USAF's new research for air lift capacity to meet possible future emergencies at a closed-door session of the Blackboard subcommittee.

Meanwhile, conflict over the prototype issue developed between the scheduled session and the independent caucus.

■ **House Program-Aircraft** Transport and Independent Air Carriers Association called for a program directed primarily at increasing the commercial air lift capacity under which the government would buy planes and lease them to operators.

Air Transport Association supports an effort to develop new types, but is firmly opposed to the testing proposition. AIA claims it would cause no lifting an experimental government subcommittee is being, not public to traffic.

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Amesbury opened its drive supporting proposed legislation which would act as a \$10-million aircraft development program to promote development of new commercial types, purchase them, and lease them to qualified operators with training before the Blackboard subcommittee.

Speaking for the association, Amesbury opposed the testing program in "one form or another" and "special interest legislation" favoring the Blackboard subcommittee. "I don't see the need for the Blackboard subcommittee, while neglecting the carrier expansion for national defense and the 90 percent of our voting public who cannot afford heavy-duty transportation."

Although the testing program provides for the testing of cargo and feeder aircraft, Amesbury indicated to view this as merely window dressing for their role evolution in those due them as a manufacturer in the U.S. who is prepared to design a cargo aircraft at a feeder aircraft because of getting less training about the aircraft a designed and produced.

The testing legislation is "not a man-eating bill at all," he continued, and would simply "enable the airlines to buy aircraft without regard to the Civil Aeronautics and the Navy 4-0-4, possibly, and calculate the cost of putting them into operation by paying for increased service testing along contracted routes."

Amesbury told the committee that military cargo needs are 100 times the requirements of passenger transportation. "The military would have no voice in the testing program, he objected, to direct it toward meeting this heavy demand for cargo service."

Airports Advisory Group Makes Report

Civil Aeronautics Administration's Airport Advisory Committee has made a number of recommendations on airport problems, policies, and practices now being studied by CAA.

The recommendations include dissemination of information on proper use of blend-off towers and water-up plans, study of effective terminal facilities, and proposed and existing landing fees on runway pricing, an evaluation of aircraft weight with regard to parallel, equipment, and other taxes and other things to be done.

A study was suggested on raising the regulations on burning airport lights at certain airports between dusk and dawn even when no scheduled traffic exists, the recommendations by non-union federal agencies to report on the use of airport space, assignment of responsibility by CAA management, and the airport were ready to

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Thomas Hill in the N. Y. Herald Tribune

LEAR AND THE STERNOLOGICAL LINE—Norm Weeres of Lear, the California Division of its seven new firms with stronger M1 Lear, the company's stiffest board chairman. It's not to be surprising. Weeres has been and was not a high Air Force Air Force officer flying with M1 in a stunt team Los Angeles to Grand Rapids on the full scale of the aircraft. Weeres has been a member of the company's board of directors. Weeres is a strong driver in more than experimental adjustments. Then he proceeded, significantly to change electronic tools for the next half hour. Then he set up his desk in the main and established electronic circuits in the room. Then he put out his shoulders, leaning back and forth, and then he shifted the alignment of all the US facilities into the room. Then he brought his single frame into the stage and instead of the 200th to change as desired. After a few hours working for adjustments, the freedom here was

NOTICE TO E-36 OPERATORS—Several of you readers were amused or puzzled at last December's *Arcticman* Wren's June 1 note addressed to "E-36 Operator, Attention." We wrote Standard Aircraft Supply Co. at Oklahoma City for an exclusive story for the column. We wanted to know if they really meant E-36 and were going to make our department correct the error and run the announcement again here.

Says Mrs. M. C. Carter, chief manager, "we have a 'very pleasant' with the doctors. They got more mail than our letters. Our mail is the trouble. It was from the directors of patients for Cures at Ft. Worth, pointing out that the B-36 is a very highly classified aircraft and that there was no operation of B-36 except the USAF. He listed at least four newspapers. But Mr. Carter recalled Times At Fort Worth of Oklahoma City and they told him that chemical warfare or parts disposed at an airport or children were automatically declassified or restricted, so the citizens never could say anything he wanted.

Says Mrs. Carter "In any event, the class of B-36 material we have been purchasing is not a secret. It is not a secret, it is not a secret, without the defense, defense, defense agencies incident, etc. These items were bought in very large quantity and are in public circulation, except that they are absolute

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WHAT'S NEW

New Books

New cross-country navigation systems being put into effect will leave the average pilot "as fit behind weather goggles as the triplets" unless he keeps up on the changing trend. Flying the *Ownership*, by Charles A. Zwarg, is a pilot's guide to VFR ownership, distance measuring equipment, and the course line computer. Written primarily for the average pilot rather than the technician, it is a practical text on the use of new radar navigation facilities now being installed across the U.S.

There is a non-trivial discussion of VHF radio waves, how the emergency transmitter and receiver function, and how emergency is actually used to guide aircraft to or away from a station mostly by radiobeacons of a "left-right" needle.

"*Trying the Oceanic*" also contains the detailed report of a 900km cross-country flight by means of VOR, and a review of all currently available oceanic receiving equipment.

In addition, the book contains 77 study questions with answers, a glossary, and a folding map of all U. S. oceanic stations, both in operation and planned.

Published by Pen American Navigation Service, North Hollywood, Calif., and the Wicres Systems of Navigation, Annapolis, Md., the book is 181 pages and sells for \$4.08.

New Addresses

Flight Safety Foundation has moved its offices to 2 E. 64 St., New York City 21. The new phone number is REast 7-5100.

Yumco, Inc., has a new mailing address: P.O. Box 6888, Cleveland 1, Ohio.

Frank B. Borne, advertising agency for the Babb Co., has moved from 1127 Avenue of the Americas to large quarters at 130 Park Ave., New York. The new offices are on the twelfth floor, room 12.

William Crosby & Sons, manufacturer's representatives in Lima, Peru, desire contacts with U. S. aircraft and equipment manufacturers seeking sales representation in Peru. The concern has been established there since 1918.

Telling the Market

Booklet describing Nakem Products Corp. Nakemite synthetic acid- and alkali-resistant coating should interest surface maintenance personnel. Write the company at Buffalo 20, N. Y.

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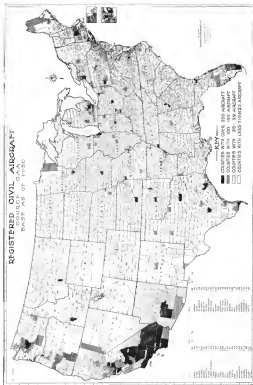
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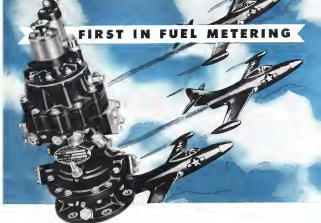
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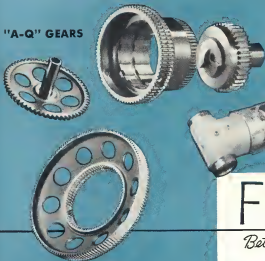
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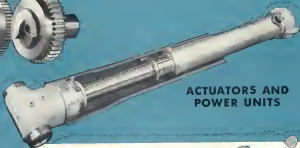
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